### Introduction

- $\triangleright$  The root locus technique shows graphically how the closed-loop poles change as a system parameter is varied.
- $\triangleright$  Used to analyze and design systems for stability and transient response.
- $\triangleright$  Shows graphically the effect of varying the gain on things like percent overshoot, and settling time.
- $\triangleright$  Also shows graphically how stable a system is; shows ranges of stability, instability, and when system will start oscillating.

### The Control System Problem

- $\triangleright$  The poles of the open-loop transfer function are typically easy to find and do not depend on the gain, *K*.
- $\triangleright$  It is thus easy to determine stability and transient response for an open-loop system.

$$
\blacktriangleright \text{ Let } G(s) = \frac{N_G(s)}{D_G(s)} \text{ and } H(s) = \frac{N_H(s)}{D_H(s)}.
$$



Figure 8.1.

### The Control System Problem - II

 $\triangleright$  Our closed transfer function is thus

$$
T(s) = \frac{K \frac{N_G(s)}{D_G(s)}}{1 + K \frac{N_G(s)}{D_G(s)} \frac{N_H(s)}{D_H(s)}}
$$
(1)

$$
=\frac{KN_G(s)D_H(s)}{D_G(s)D_H(s)+KN_G(s)N_H(s)}\tag{2}
$$

 $\blacktriangleright$  We thus see that we have to factor the denominator of  $T(s)$ to find the closed-loop poles, and they will be a function of *K*.

$$
\begin{array}{c|c}\n\hline\nR(s) & KG(s) & C(s) \\
\hline\n1 + KG(s)H(s)\n\end{array}
$$



#### The Control System Problem - III

• For example, if 
$$
G(s) = \frac{s+1}{s(s+2)}
$$
 and  $H(s) = \frac{s+3}{s+4}$ , our closed-loop transfer function is:

$$
T(s) = \frac{K(s+1)(s+4)}{s(s+2)(s+4) + K(s+1)(s+3)}
$$
(3)

$$
=\frac{K(s+1)(s+4)}{s^3+(6+K)s^2+(8+4K)s+3K}
$$
 (4)

- $\triangleright$  To find the poles, we would have to factor the polynomial for a specific value of *K*.
- $\triangleright$  The root-locus will give us a picture of how the poles will vary with *K*.

#### Vector Representation of Complex Numbers

- Any complex number,  $\sigma + j\omega$ , can be represented as a vector.
- It can be represented in polar form with magnitude  $M$ , and an angle  $\theta$ , as  $M\angle\theta$ .
- If  $F(s)$  is a complex function, setting  $s = \sigma + i\omega$  produces a complex number. For  $F(s) = (s + a)$ , we would get  $(\sigma + a) + j\omega$ .



Figure 8.2.

#### Vector Representation of Complex Numbers - II

- If we note that function  $F(s) = (s + a)$  has a zero at  $s = -a$ , we can alternately represent  $F(\sigma + j\omega)$  as originating at  $s = -a$ , and terminating at  $\sigma + i\omega$ .
- $\triangleright$  To multiply and divide the polar form complex numbers,  $z_1 = M_1\angle\theta_1$  and  $z_2 = M_2\angle\theta_2$ , we get

$$
z_1 z_2 = M_1 M_2 \angle (\theta_1 + \theta_2) \qquad \frac{z_1}{z_2} = \frac{M_1}{M_2} \angle (\theta_1 - \theta_2) \qquad (5)
$$



Figure 8.2.

#### Polar Form and Transfer Functions

 $\blacktriangleright$  For a transfer function, we have:

<span id="page-6-0"></span>
$$
G(s) = \frac{(s+z_1)\cdots(s+z_m)}{(s+p_1)\cdots(s+p_n)} = \frac{\prod_{i=1}^m (s+z_i)}{\prod_{i=1}^n (s+p_i)} = M_G \angle \theta_G
$$
\n(6)

where

$$
M_G = \frac{\prod_{i=1}^{m} |(s+z_i)|}{\prod_{i=1}^{n} |(s+p_i)|} = \frac{\prod_{i=1}^{m} M_{z_i}}{\prod_{i=1}^{n} M_{p_i}}
$$
(7)

and

$$
\theta_G = \sum \text{zero angles} - \sum \text{pole angles} \tag{8}
$$
\n
$$
= \sum_{i=1}^{m} \angle (s + z_i) - \sum_{j=1}^{n} \angle (s + p_j) \tag{9}
$$

## Polar Form and Transfer Functions eg.

• Use Equation 6 to evaluate 
$$
F(s) = \frac{(s+1)}{s(s+2)}
$$
 at  $s = -3 + j4$ .



Figure 8.3.

# Polar Form and Transfer Functions eg.



Figure 8.3.

Check answer using matlab:

 $1$  s=tf('s') 2  $F= (s+1)/(s*(s+2))$ <br>3  $s1=evalfr(F,-3+i*4)$  $s1 = e \nu a$  If  $r(F, -3 + i * 4)$ 4  $M=abs(s1)$  $5$  theta=angle (s1)  $6$  theta  $*180/pi$ c 2006-2012,2017 R.J. Leduc & M. Lawford 10

# Root Locus Introduction

- $\triangleright$  System below can automatically track subject wearing infrared sensors.
- $\triangleright$  Solving for the poles using the quadratic equation, we can create the table below for different values of *K*.



 $(a)$ 



Table 8.1.

κ	Pole 1	Pole 2
$\mathbf{0}$	$-10$	0
5	$-9.47$	$-0.53$
10	$-8.87$	$-1.13$
15	$-8.16$	$-1.84$
20	$-7.24$	$-2.76$
25	$-5$	$-5$
30	$-5 + i2.24$	$-5 - i2.24$
35	$-5 + j3.16$	$-5 - i3.16$
40	$-5 + j3.87$	$-5 - j3.87$
45	$-5 + i4.47$	$-5 - j4.47$
50	$-5 + i5$	$-5 - i5$

 $\widehat{\mathsf{c}}$  2006-2012,2017 R.J. Leduc & M. Lawford  $\qquad \qquad \qquad \qquad \qquad \qquad \mathsf{Figure:} \ \ 8.4$ 

# Root Locus Introduction - II

 $\triangleright$  We can plot the poles from Table 8.1. labelled by their corresponding gain.

Table 8.1.





Figure: 8.5

## Root Locus Introduction - III

- $\triangleright$  We can go a step further, and replace the individual poles with their paths.
- $\triangleright$  We refer to this graphical representation of the path of the poles as we vary the gain, as the root locus.
- $\blacktriangleright$  We will focus our discussion on  $K \geq 0$ .

For pole 
$$
\sigma_D + j\omega_D
$$
,  $T_s = \frac{4}{\sigma_D}$ ,  $T_p = \frac{\pi}{\omega_D}$ , and  $\zeta = \frac{|\sigma_D|}{\omega_n}$ .



#### Root Locus Properties

- $\triangleright$  For second-order systems, we can easily factor a system and draw the root locus.
- $\triangleright$  We do not want to have to factor for higher-order systems (5th, 10th etc.) for multiple values of *K*!
- $\triangleright$  We will develop properties of the root locus that will allow us to rapidly sketch the root locus of higher-order systems.
- $\triangleright$  Consider the closed-loop transfer function below:

$$
T(s) = \frac{KG(s)}{1 + KG(s)H(s)}
$$

 $\blacktriangleright$  A pole of  $T(s)$  exists when

<span id="page-12-0"></span>
$$
KG(s)H(s) = -1 = 1 \angle (2k+1)180^o \qquad k = 0, \pm 1, \pm 2, \dots
$$
\n(10)

#### Root Locus Properties - II

 $\blacktriangleright$  Equation [10](#page-12-0) is equivalent to

<span id="page-13-1"></span>
$$
|KG(s)H(s)| = 1 \tag{11}
$$

and

<span id="page-13-0"></span>
$$
\angle KG(s)H(s) = (2k+1)180^o \tag{12}
$$

- $\blacktriangleright$  Equation [12](#page-13-0) says that any  $s'$  that makes the angle of  $KG(s)H(s)$  be an odd multiple of  $180^{\circ}$  is a pole for some value of *K*.
- $\triangleright$  Given *s'* above, the value of K that *s'* is a pole of  $T(s)$  for is found from Equation [11](#page-13-1) as follows:

$$
K = \frac{1}{|G(s)||H(s)|}
$$
(13)

#### Root Locus Properties eg.

For system below, consider  $s = -2 + j3$  and  $s = -2 + j(\sqrt{2}/2).$ 



Figures 8.6 and 8.7.

# Sketching Root Locus

- $\triangleright$  Now give a set of rules so that we can quickly sketch a root locus, and then we can calculate exactly just those points of particular interest.
	- 1. Number of branches: a branch is the path a single pole traverses. *The number of branches thus equals the number of poles.*
	- 2. Symmetry: As complex poles occur in conjugate pairs, *a root locus must be symmetric about the real axis*.



# Sketching Root Locus - II

3. Real-axis segments: *For K >* 0*, the root locus only exists on the real axis to the left of an odd number of finite open-loop poles and/or zeros, that are also on the real axis.*

Why? By Equation [12,](#page-13-0) the angles must add up to an odd multiple of 180.

- $\triangleright$  A complex conjugate pair of open-loop zeros or poles will contribute zero to this angle.
- $\triangleright$  An open-loop pole or zero on the real axis, but to the left of the respective point, contributes zero to the angle.
- $\triangleright$  The number must be odd, so they add to an odd multiple of 180, not an even one.



Figure 8.8.

### Sketching Root Locus - III

4. Starting and ending points: *The root locus begins at the finite and infinite poles of G*(*s*)*H*(*s*) *and ends at the finite and infinite zeros of*  $G(s)H(s)$ .

Why? Consider the transfer function below

$$
T(s) = \frac{KN_G(s)D_H(s)}{D_G(s)D_H(s) + KN_G(s)N_H(s)}
$$

 $\triangleright$  The root locus begins at zero gain, thus for small  $K$ , our denominator is

$$
D_G(s)D_H(s) + \epsilon \tag{14}
$$

 $\blacktriangleright$  The root locus ends as  $K$  approaches infinity, thus our denominator becomes

$$
\epsilon + KN_G(s)N_H(s)
$$

### Infinite Poles and Zeros

 $\triangleright$  Consider the open-loop transfer function below

$$
KG(s)H(s) = \frac{K}{s(s+1)(s+2)}
$$
(15)

- ▶ From *point 4*, we would expect our three poles to terminate at three zeros, but there are no finite zeros.
- $\triangleright$  A function can have an infinite zero if the function approaches zero as *s* approaches infinity. ie.  $G(s) = \frac{1}{s}$ .
- $\triangleright$  A function can have an infinite pole if the function approaches infinity as *s* approaches infinity. ie.  $G(s) = s$ .
- $\triangleright$  When we include infinite poles and zeros, every function has an equal number of poles and zeros

$$
\lim_{s \to \infty} KG(s) H(s) = \lim_{s \to \infty} \frac{K}{s(s+1)(s+2)} \approx \frac{K}{s \cdot s \cdot s} \quad (16)
$$

How do we locate where these zeros at infinity are so we can terminate our root locus?

 $\odot$ 2006-2012.2017 R.J. Leduc & M. Lawford  $20$ 

## Sketching Root Locus - IV

5. Behavior at Infinity: *As the locus approaches infinity, it approaches straight lines as asymptotes*.

*The asymptotes intersect the real-axis at*  $\sigma_a$ , and depart at *angles*  $\theta_a$ *, as follows:* 

$$
\sigma_a = \frac{\sum \text{finite poles} - \sum \text{finite zeros}}{\# \text{finite poles} - \# \text{finite zeros}} \tag{17}
$$
\n
$$
\theta_a = \frac{(2k+1)\pi}{\# \text{finite poles} - \# \text{finite zeros}} \tag{18}
$$

*where*  $k = 0, \pm 1, \pm 2, \pm 3$ , and the angle is in radians relative *to the positive real axis.*

# Sketching Root Locus eg. 1

 $\triangleright$  Sketch the root locus for system below.



Figure 8.11.

## Real-axis Breakaway and Break-in Points

▶ Consider root locus below

- $\triangleright$  We want to be able to calculate at what points on the real axis does the locus leave the real-axis (breakaway point), and at what point we return to the real-axis (break-in point).
- $\triangleright$  At breakaway/break-in points, the branches form an angle of  $180^\circ/n$  with the real axis where *n* is number of poles converging on the point.



Figure 8.13.

## Real-axis Breakaway and Break-in Points - II

- $\triangleright$  Breakaway points occur at maximums in the gain for that part of the real-axis.
- $\triangleright$  Break-in points occur at minimums in the gain for that part of the real-axis.
- $\triangleright$  We can thus determine the breakaway and break-in points by setting  $s = \sigma$ , and setting the derivative of equation below equal to zero:



#### Real-axis Breakaway and Break-in Points - III

An alternative method for computing the real-axis breakaway and break-in points without differentiation is to solve the equation:

$$
\sum_{i=1}^{m} \frac{1}{\sigma + z_i} = \sum_{i=1}^{n} \frac{1}{\sigma + p_i}
$$
 (8.37)

where  $z_i$  and  $p_i$  are the negative of the zeros and poles, respectively, of  $G(s)H(s)$ . i.e.

$$
G(s)H(s) = \frac{K_{GH}(s+z_1)(s+z_2)\dots(s+z_m)}{(s+p_1)(s+p_2)\dots(s+p_n)}
$$

# The *jw*-Axis Crossings

- $\triangleright$  For systems like the one below, finding the  $j\omega$ -axis crossing is important as it is the value of the gain where the system goes from stable to unstable.
- $\triangleright$  Can use the Routh-Hurwitz criteria to find crossing:
	- 1. Force a row of zeros to get gain
	- **2.** Determine polynomial for row above to get  $\omega$ , the frequency of oscillation.



Figure 8.12.

# The  $j\omega$ -Axis Crossing eg.

 $\triangleright$  For system below, find the frequency and gain for which the system crosses the  $j\omega$ -axis.



Figures 8.11 and 8.12.

# Angles of Departure and Arrival

- $\triangleright$  We can refine our sketch by determining at which angles we depart from complex poles, and arrive at complex zeros.
- $\triangleright$  Net angle from all open-loop poles and zeros to a point on root access must satisfy:

<span id="page-26-0"></span>
$$
\Sigma zero \ angles - \Sigma pole \ angles = (2k + 1)180^o \qquad (20)
$$

 $\triangleright$  To find angle  $\theta_1$ , we choose a point  $\epsilon$  on root locus near complex pole, and assume all angles except  $\theta_1$  are to the complex pole instead of  $\epsilon$ . Can then use Equation [20](#page-26-0) to solve for  $\theta_1$ .



## Angles of Departure and Arrival - II

► For example in Figure 8.15a, we can solve for  $\theta_1$  in equation below:

<span id="page-27-0"></span>
$$
\theta_2 + \theta_3 + \theta_6 - (\theta_1 + \theta_4 + \theta_5) = (2k + 1)180^{\circ} \tag{21}
$$

- $\triangleright$  Similar approach can be used to find angle of arrival of complex zero in figure below.
- Simply solve for  $\theta_2$  in Equation [21.](#page-27-0)



## Angles of Departure and Arrival eg.

 $\blacktriangleright$  Find angle of departure for complex poles, and sketch root locus.



Figures 8.16 and 8.17.

## Plotting and Calibrating Root Locus

- $\triangleright$  Once sketched, we may wish to accurately locate certain points and their associated gain.
- $\triangleright$  For example, we may wish to determine the exact point the locus crosses the 0.45 damping ratio line in figure below.

► From Figure 4.17, we see that 
$$
cos(\theta) = \frac{adj}{hyp} = \frac{\zeta \omega_n}{\omega_n} = \zeta
$$
.

 $\blacktriangleright$  We then use computer program to try sample radiuses, calculate the value of *s* at that point, and then test if point satisfies angle requirement.



 $\odot$ 2006-2012,2017 R.J. Leduc & M. Lawford  $^{Figures~4.17}$  and 8.18.  $^{31}$ 

## Plotting and Calibrating Root Locus - II

 $\triangleright$  Once we have found our point we can use the equation below to solve for the required gain, *K*.

$$
K = \frac{1}{|G(s)||H(s)|} = \frac{\prod_{i=1}^{m} M_{p_i}}{\prod_{i=1}^{n} M_{z_i}}
$$
(22)

 $\triangleright$  Uses labels in Figure 8.18, we would have for our example:

$$
K = \frac{ACDE}{B} \tag{23}
$$



Figures 4.17 and 8.18.

#### Transient Response Design via Gain Adjustment

- $\triangleright$  We want to be able to apply our transient response parameters and equations for second-order underdamped systems to our root locuses.
- $\blacktriangleright$  These are only accurate for second-order systems with no finite zeros, or systems that can be approximated by them.
- $\triangleright$  What are the conditions that must be true for a 2nd order approxiation to be "close" to the higher order system?

Recall if 
$$
G(s) = \frac{N_G(s)}{D_G(s)}
$$
 and  $H(s) = \frac{N_H(s)}{D_H(s)}$ , closed loop TF is:

$$
T(s) = \frac{KN_G(s)D_H(s)}{D_G(s)D_H(s) + KN_G(s)N_H(s)}
$$
(24)

# Transient Response Design via Gain Adjustment II

Conditions for justifying 2*nd* order approx of a higher order systems

- 1. Higher order poles are much farther left (e.g.  $> 5\times$ ) of the *s*-plane dominant closed loop poles. (Holds for (b), not (a))
- 2. The closed-loop zeros near the two dominant closed-loop poles must be nearly canceled by higher-order poles near them. (Holds for (d), not (c))
- 3. Closed-loop zeros not cancelled, must be far away from the two dominant closed-loop poles.



#### Defining Parameters on Root Locus

- $\triangleright$  We have already seen that as  $\zeta = \cos \theta$ , vectors from the origin are lines of constant damping ratio.
- As percent overshoot is solely a function of  $\zeta$ , these lines are also lines of constant %OS.
- $\triangleright$  From diagram we can see that the real part of a pole is  $\sigma_d = \zeta \omega_n$ , and the imaginary part is  $\omega_d = \omega_n \sqrt{1 - \zeta^2}$ .



#### Defining Parameters on Root Locus - II

• As 
$$
T_p = \frac{\pi}{\omega_n \sqrt{1 - \zeta^2}} = \frac{\pi}{\omega_d}
$$
, horizontal lines thus have constant peak time.

 $\triangleright$  We thus choose a line with the desired property, and test to find where it intersects our root locus.



# Design Procedure For Higher-order Systems

- 1. Sketch root locus for system.
- 2. Assume system has no zeros and is second-order. Find gain that gives desired transient response.
- 3. Check that systems satisfies criteria to justify our approximation.
- 4. Simulate system to make sure transient response is acceptable.

### Third-order System Gain Design eg.

- $\triangleright$  For system below, design the value of gain,  $K$ , that will give 1.52% overshoot. Also estimate the settling time, peak time, and steady-state error.
- $\blacktriangleright$  First step is to sketch the root locus below.
- $\triangleright$  We next assume system can be approximated by second-order system, and solve for  $\zeta$  using  $\zeta = \frac{-\ln(\mathcal{V}_0OS/100)}{\sqrt{\pi^2+\ln^2(\mathcal{V}_0OS/100)}}$ .  $\zeta = 0.8$  $i4$  $-4.6 + j3.45$ ,  $K = 39.64$  $j3$ s-plane  $i2$  $-1.19 + i0.90$ ,  $K = 12.79$  $C(s)$  $E(s)$  $-0.87 + j0.66$ ,  $K = 7.36$  $(s + 1)(s + 10)$ - 0  $-4$  $-3$  $-1.5 \mathbf{0}$ -71  $X = Closed-loop pole$  $X =$ Open-loop pole -12

# Third-order System Gain Design eg. - II

- **Figure** This gives  $\zeta = 0.8$ . Our angle is thus  $\theta = \cos^{-1}(0.8)$  $= 36.87$ <sup>o</sup>
- $\triangleright$  We then use root locus to search values along this line to see if they satisfy the angle requirement.
- $\triangleright$  The program finds three conjugate pairs on the locus and our  $\zeta = 0.8$  line. They are  $-0.87 \pm j0.66$ ,  $-1.19 \pm j0.90$ ,  $-4.6 \pm j3.45$  with respective gains of  $K = 7.36, 12.79$ , and 39*.*64.



Figures 8.21 and 8.22.

# Third-order System Gain Design eg. - III

 $\blacktriangleright$  For steady-state error, we have:

$$
K_v = \lim_{s \to 0} sG(s) = \lim_{s \to 0} s \frac{K(s+1.5)}{s(s+1)(s+10)} = \frac{K(1.5)}{(1)(10)} \tag{25}
$$

- $\triangleright$  To test to see if our approximation of a second-order system is valid, we calculate the location of the third pole for each value of *K* we found.
- $\triangleright$  The table below shows the results of our calculations.



### Third-order System Gain Design eg. - IV

 $\triangleright$  We now simulate to see how good our result is:



Figure 8.23.